V 1.16 of MEMS integrated inertial navigation system.





Product characteristics



(GNSS Effective) 0.1 ° Roll & Pitch Attitude Accuracy



0.1 ° azimuth accuracy (2m antenna baseline)



2°/H gyro bias stability (Allan)



30 μg acceleration bias stability (Allan)

Field of application

UAV Navigation Robot Navigation AUV Navigation

Various air carriers flight navigation land vehicle navigation ROV navigation







1-General

BS-MN100B-M-D6EC integrated navigation system is composed of MEMS sensors and high-end GNSS receiver board, which is realized by multi-sensor fusion and navigation algorithm. The product has high reliability and strong environmental adaptability. By matching different software, the product can be widely used in UAV, UAV, surveying and mapping, marine compass, stable platform, underwater vehicle and other fields.

2-Functions and indicators

2.1 Main functions

The integrated navigation system can use the satellite navigation information received by the GNSS receiver to carry out integrated navigation, and output the information of the carrier, such as pitch, roll, course, position, speed, time and the like; After losing the signal, it outputs the position, velocity and attitude information of inertial solution, and has a certain navigation accuracy maintenance function in a short time. When combined with navigation, it can output the raw information that can be used for post-processing to be processed by the IE post-processing software of NovAtel.

2.2 Performance indicators

The performance of integrated navigation system is shown in Table 1.

Table 1 System Performance Index

Project	Metrics (RMS)	Remark

Pro	oject	Metrics (RMS)	Remark
		Heading accuracy	
	Dual GNSS	0.1°	2m baseline
	Single GNSS	0.2°	Need to maneuver
	Post-processing	0.03°	Optional
	Maintain accuracy	1°/min	GNSS failure
		Attitude accuracy	
	GNSS is valid	0.1°	Single-point L1/L2,
	Inertial/odometer	0.1°	Optional
	Post-processing	0.02°	Optional
	Maintain accuracy	1°/min	GNSS failure
Horizontal positioning accuracy			
	GNSS is valid	1.2m	Single-point L1/L2
	Inertial/odometer	2 ‰ D (D represents	Optional
	Post-processing	1cm+1ppm	Optional
	GNSS failure	1m/10s、10m/30s、20m/60	s (RMS)
	Hor	izontal velocity accuracy	
	GNSS is valid	0.1m/s	Single-point L1/L2,
	Inertial/odometer	0.1m/s (RMS)	Optional
	Inertial/DVL combination	0.2m/s (RMS)	Optional
		Gyroscope	
	Measuring range	±450°/s	
	Zero bias stability	2°/h	Allan variance
		Accelerometer	
	Measuring range	±16g	Customizable 200 G
	Zero bias stability	30μg	Allan variance
		Satellite card	
	Cold start	30 seconds	Open environment
	Hot start	5 seconds	Open environment
	Type of receiver	BDS B1I/B2I/B3I	

Project		Metrics (RMS)	Remark		
	Co	mmunication interface			
	RS232&RS422	Route 7			
	CAN	Route 1	Optional		
	Odometer differential input	Route 1	Optional		
	PPS output	Route 1	Optional		
	El	ectrical characteristics			
	Voltage	12~36VDC			
	Power consumption	≤3W			
	Ripple	100 mV (P-P)			
	St	ructural characteristics			
	Size	80 mm×53 mm×23mm			
	Weight	≤150g			
		Use environment			
	Operating temperature	-40°C~+60°C			
	Storage temperature	-45°C~+65°C			
	Vibration	20~2000Hz, 6.06g			
	Impact	30g, 11ms			
	Degree of protection	IP65			
	Reliability				
	MTBF	30000h			
	Life span	> 15 years			
	Continuous working time	>24h			

3. Working principle

3.1 Product composition

The composition of the product is shown in Figure 1.

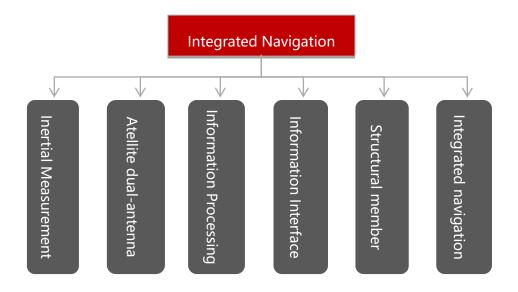


Figure 1 System composition

3.2 Fundamentals

The inertial measurement unit consists of three accelerometers and three gyroscopes and is used for measuring the acceleration and the angular velocity of a carrier and sending the information to the information processing circuit; and the information processing circuit performs navigation settlement by using the acceleration and the angular velocity measured by the inertial measurement unit and simultaneously receives satellite navigation information of a GNSS receiver as a reference to perform combined navigation, The navigation error of the inertial navigation is corrected, and the navigation information is output through the information interface circuit.

The basic principle is shown in Figure 2.

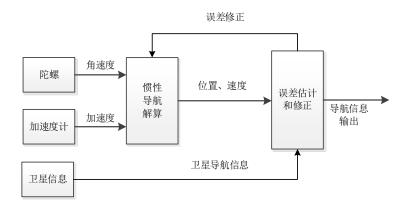


Fig. 2 Schematic diagram of working principle

4.Instructions for use

4.1 overall dimensions

The inertial measurement unit and GNSS receiver adopt an integrated design scheme and are integrated in the integrated navigation system. The system outline is shown in Figure 3.

Overall dimension of integrated navigation system: 117mm \times 96mm \times 39.8 mm (length \times width \times height).

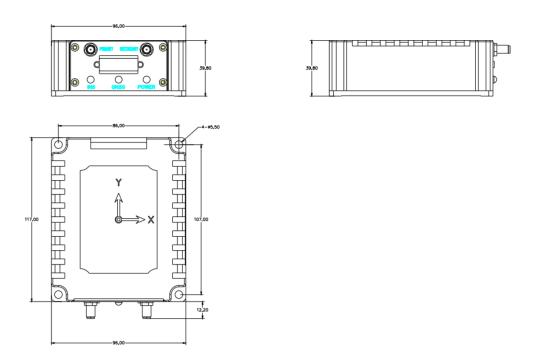


Fig. 3 Outline structure of integrated navigation system

4.2 Electrical interface

4.2.1 product interface

The system has 3 external connectors:

One communication interface, two RF cable interfaces (SMA outer thread and inner hole, where the left is connected to the main antenna and the right is connected to the auxiliary antenna, facing the interface definition on the left and right), one end of the communication cable is a J30J connector, which is connected to the system, and the other end is divided into a power line and a communication line (except that the network cable is a 0.5 m, the Other cable is 1.5 m):

Power cable: connected to $9 \sim 36$ V DC, the working current is not more than 0.4 A when the power supply is 24 V, the external wire is thrown, and the red and black clips are connected.

Communication cable: with 7 serial ports, 1 Ethernet interface, 1 CAN interface, PPS interface, EVENT interface and odometer interface. COM1 is used to send working mode instructions, which is RS232; COM2, COM3 and COM4 are protocol output interfaces, which are RS232 and RS422 configurable; COM5 is used for product debugging, which is RS232; GNSS _ COM 1 and GNSS _ COM3 are directly connected to COM1 and COM3 of GNSS receiver board; Serial ports are all BD 9 female. CAN interface, PPS interface, EVENT interface and odometer interface are connected out in a line-throwing mode.

Two RF cables: one end is connected to the antenna, and the other end is connected to the product.

Table 2 J30J (J30J-51ZKWP7-J) Connector Point Definition

J30J	Signal definition	Identification	Handling	Remark
1	IN1	IN1+	Throw the	The first road
2	IN1_GND	IN1-	Throw the	

J30J	Signal definition	Identification	Handling	Remark	
3	IN2	IN2+	Throw the	The second	
4	IN2_GND	IN2-	Throw the		
6	GNSS_EVENT2	617_EVENT2	Throw the		
7	GNSS_PPS	617_PPS	Throw the		
8	GNSS_GND	GND	Throw the		
9	COM1_RXD		1-DB 9 holes:		
10	COM1_TXD	COM1	1-DB 9 holes:	RS232	
11	COM1_GND		1-DB 9 holes:		
14	COM2_RS422_TX		2-DB 9 holes:	DC 400 (DC000	
15	COM2_RS422_TX		2-DB 9 holes:	RS422/RS232	
17	COM2_RS422_RX	COM2	2-DB 9 holes:	Default RS	
16	COM2_RS422_RX		2-DB 9 holes:	232	
18	COM2_GND		2-DB 9 holes:		
35	COM3_RS422_TX		6-DB 9 holes:	DC422/DC222	
34	COM3_RS422_TX		6-DB 9 holes:	RS422/RS232	
51	COM3_RS422_RX	COM3	6-DB 9 holes:	RS422 is the	
50	COM3_RS422_RX		6-DB 9 holes:	default	
36	COM3_GND		6-DB 9 holes:		
48	COM4_RS422_TX		7-DB 9 holes:	DC 400 (DC000	
47	COM4_RS422_TX		7-DB 9 holes:	RS422/RS232	
46	COM4_RS422_RX	COM4	7-DB 9 holes:	RS422 is the	
45	COM4_RS422_RX		7-DB 9 holes:	default	
44	COM4_GND		7-DB 9 holes:		
19	24V GND	24V CND	Black		
20	24V_GND	24V_GND	III	Dawen susseli	
21	24V	241/	Red alligator	Power supply	
22	24V	24V	<u> </u>		
			•		

J30J	Signal definition	Identification	Handling	Remark	
23	GNSS_COM3_RX		3-DB 9 holes:		
24	GNSS_COM3_TX	GNSS_COM3	3-DB 9 holes:	RS232	
25	GNSS_COM3_GN		3-DB 9 holes:		
26	GNSS COM1 TX		4-DB 9 holes:		
27	GNSS COM1 RX	GNSS COM1	4-DB 9 holes:	RS232	
28	GNSS COM1 GN	_	4-DB 9 holes:		
20	GN33_COM1_GN		4-DB 9 Holes.		
29	TEST_COM_RS232		5-DB 9 holes:	Debugging	
30	TEST_COM_RS232	COM5	5-DB 9 holes:		
31	TEST_COM_GND		5-DB 9 holes:	port, RS232	
37	ETHER_RX_N (2)		RJ45-6(NET)		
38	ETHER_RX_P (1)	Ethernet	RJ45-3(NET)	10 M Ethernet	
39	ETHER_TX_N (6)	201011100	RJ45-2(NET)	interface	
40	ETHER_TX_P (3)		RJ45-1(NET)		
41	CAN GND		Throw the		
42	CAN P	CAN	Throw the	CAN	
	_				
43	CAN_N		Throw the		
49	GND				

4.3 Instructions for use

4.3.1.1startup prompt message

Connect the cable, power on the system, monitor the COM1 interface information through the serial port debugging tool of the test computer, and display "" on the interface for Please enter NaviMode within 20s! After "", workflow instructions can be sent to the COM1 interface through the serial port debugging tool. "# moddgi" is the work instruction to enter the integrated navigation; "# modins" is the work

instruction to enter the inertial navigation. If no command is sent within 20 s, the system will automatically enter the integrated navigation workflow after 20 s. After entering the integrated navigation process, first bind the satellite information. If the satellite is not positioned, it is in the state of waiting for the satellite information. When the satellite information is valid, it enters the alignment state. The alignment time is 10 s. During the alignment period, the integrated navigation system is required to be static. After the alignment is completed, the integrated navigation system can move, and the system is in an integrated navigation state. After entering the inertial navigation process, the satellite information is bound first. If the satellite is not positioned, it is in the state of waiting for the satellite information. When the satellite information is valid, it enters the coarse alignment state. The coarse alignment time is 10 s. During the coarse alignment, the integrated navigation system is required to be static. After the coarse alignment is completed, the integrated navigation system can move and the system is in the fine alignment state. After fine alignment for 1500 s, it will automatically switch to the inertial navigation state.

The 4.3.1.2 system is reset

During operation, input the "# reset" command, and the system will perform soft reset and display the startup information again.

4.3.1.3 Store Data Export

This product is equipped with optional data storage function. If it is equipped with data storage function, the total storage space is 16G. After the system completes

the coarse alignment, the system automatically saves the storage information according to the user's configuration. The name of the saved data folder is recordX, where X is the file number (up to 39), and the number increases in turn. When X is 39, record00 will be automatically overwritten in the next storage, and X will still increase in turn in the next storage. If the system is configured to store data, the system will automatically delete the oldest data folder after each power-on operation. For example, if the currently generated file is record08, there will be no record09 folder in the system memory. Users can use this as a basis to find the latest data file.

The recordX folder contains various protocol files configured by the user. Each protocol is a separate file, and the file name is the protocol name.

The data export operation is as follows:

Connect the Ethernet interface to the test computer;

Tting the IP address of the test computer as a 192.168.1.22;

Power on the system;

Run SSH software (SSH Secure File Transfer Client);

Click Quick Connect and fill in the contents as shown in Figure 4;

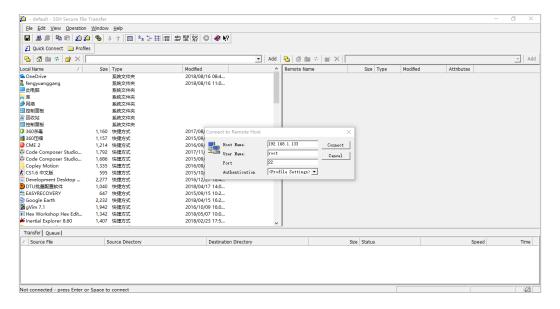


Figure 4 SSH Software Connection Diagram

Click Connect, and the SSH software will automatically pop up the "Enter Password" dialog box. There is no need to fill in the password. Click OK to connect to the system;

Enter/media/mmcblk0p1 in the address bar on the right side of the figure above and press Enter. Find the corresponding storage directory in the directory below to download.

When the download operation is complete, the system is powered off and the Ethernet interface is unplugged.

4.3.2 system configuration instruction

4.3.2.1 Configuration Scheme and Storage

The integrated navigation system provides 6-channel serial port and internal storage, and the function allocation and relevant configuration of each serial port and internal storage are shown in Table 3.

String	Enter the project	Output items	Default
slogan			
s			
COM1	1. working mode	1.inspvasa bdfpd bdfpdl	115200bps;
	instruction and	bdfpdb、gpfpd、INStest(0.2Hz、	Output: bdfpd
	flow control	1Hz、5Hz、10Hz、100Hz 200Hz,	1Hz;
	instruction;	etc.);	
	2. COM1 ~ COM4	2. rawdata (200Hz 16488	
	baud rate, protocol	format);	
	and update rate	3.rawimusb、INSpost(200Hz);	
	configuration;	4.bestposa 、 gprmc 、 gpgga	
	3.COM2 ~ COM4	(1Hz);	
	serial port type	5.Configure the prompt	
	configuration	message.	
	(RS422/RS232);		
СОМ2	None	Same as items 1-4 in COM 1	460800bps
			Output: INSpost
сомз	None	Same as items 1-4 in COM 1	460800bps
			Output: rawdata
СОМ4	None	Same as items 1-4 in COM 1	460800bps
			Output: None

File	None	Same as items 1-4 in COM 1	Output: rawdata
GNSS_	GNSS COM1	The NovAtel-718D supports	9600bps;
COM1	Configuration Item	protocols and configures	Output: None
		outputs based on this interface,	
		such as bestposa, gprmc, and so	
		on	
GNSS_	GNSS COM3	The NovAtel-718D supports	115200bps;
сомз	Configuration Item	protocols and configures	Output: None
		outputs based on this interface,	
		such as bestposa, gprmc, and so	
		on	

After the system is powered on and the start prompt information is displayed on the COM1 port, you can input commands such as COM1 ~ COM4 serial port baud rate configuration, serial port protocol and update rate setting. If each command is output successfully, it will return to the "cmd OK" ", otherwise the cmd error" will be displayed. After the input is completed, type "saveconfig" to save the current configuration. The current configuration will be called automatically after the next

Table 3 Serial port function distribution of integrated navigation system

4.3.2.2 configuration query

the last saved configuration after the next restart.

Type the "log loglist" or "log rxstatus" command through the COM1 port to list all

restart. If the command is not input, the serial port configuration will be restored to

the configurations of COM1-COM4, including the following contents:

Serial port number, serial port baud rate, serial port protocol and update rate;

Open state of function module: including zero-speed correction state and smooth

processing state, enable when open and disable when closed;

Initial binding longitude and latitude;

Initially binding the included angle between the double-antenna heading and the

integrated navigation system heading;

Initial binding antenna mast arm value;

System number and date of manufacture;

Software version number: including pre-processing software version number and

navigation software version number;

Operating mode: including integrated navigation (DGI) and pure inertial navigation

4.3.2.3 baud rate configuration

In this mode, enter the following command to enter the serial port baud rate configuration:

com comX BAUDRATA

(INS).

Where X is from 1 to 4, and BAUDRATA is the baud rate in bps.

For example, set the baud rate of the COM1 port to 115200 bps, and input the following command:

com com1 115200

4.3.2.4 protocol and update rate configuration

4.3.2.4.1 protocol and update rate configuration

Configure the output protocols of COM1 ~ COM4 and the internal storage file through COM1. The configuration commands are as follows:

log comX/fileLOG ontime updataTime

Where comX can be the configuration number of com 1 to com 4, and file is the configuration number of the memory interface; The updataTime represents the update time, which can be a period of 5 (0.2 Hz), 1 (1 Hz), 0.2 (5 Hz), 0.1 (10 Hz), 0.01 (100 Hz), etc., which can be divided by 200 Hz, and the unit is s.

LOG indicates the protocol name, which can be inspvasa, bdfpd, gpfpd, etc.

For example, if you want to configure the COM2 port to output 10Hz bdfpd data, you can input the following command through COM1:

log com2 bdfpd ontime 0.1

If 10Hz inspvasa data needs to be output at COM2 at the same time, the following command can be input through COM1:

log com2 inspvasa ontime 0.1

As another example, to store inspvasa protocol data at 1 Hz to internal storage, enter the following command from COM1:

log file inspvasa ontime 1

If you want to shut down a protocol, the configuration command is as follows:

log comX/fileLOG off

Configure the rawdata protocol of the COM1-COM4 ports and the internal storage file through COM1. The configuration commands are as follows:

log comX/filerawdata onchanged

If you want to close the rawdata protocol of the serial port, the configuration command is as follows:

log comX/filerawdata off

If you want to close all protocols of the serial port, the configuration command is as follows:

unlogall comX/file

It should be noted that increasing the update rate or outputting multiple protocols at the same time will increase the amount of data sent by the serial port. Before use, it is necessary to configure the appropriate baud rate, otherwise it may cause data loss. In general, the larger the amount of data, the higher the baud rate required.

4.3.2.4.2 protocol format

The output protocols supported by the product are shown in the following table.

Table 4 Output Data Protocol Description

Serial	Data protocol	Type of	Output type	Support interface
	gpfpd	ASCII	ontime	COM1-COM4, file
	bdfpd	ASCII	ontime	COM1-COM4, file
	bdfpdb	Binary	ontime	COM1-COM4, file
	rawimusb	Binary	onchanged	COM1-COM4, file
	inspvasa	ASCII	ontime	COM1-COM4, file
	INStest	Binary	ontime	COM1-COM4, file
	bestposa	ASCII	onchanged	COM1-COM4, file
	gprmc	ASCII	onchanged	COM1-COM4, file
	gpgga	ASCII	onchanged	COM1-COM4, file
	INSpost	Binary	onchanged	COM1-COM4, file

rawdata	Binary	onchanged	COM1-COM4, file
bdfpdl	ASCII	ontime	COM1-COM4

The ASCII type protocol conforms to the NMEA protocol format requirement and comprises the following fields: a statement identifier, a plurality of data fields, Checksum End tag (with carriage return < CR > and Line break < LF >) separated by commas. Take the bdfpd protocol as an example, the format is as follows:

The rawdata protocol includes rawimusb, rangecmpb, bestvelb, bestposb, headingb, and psrdopb. The contents of rangecmpb, bestvelb, bestposb, headingb and psrdopb protocols are shown in the NovAtel protocol description. The gprmc and gpgga protocols are in NMEA 0183 protocol data format, and bestposa is in NovAtel protocol data format, both of which forward the output of the satellite board and will not be introduced here. The protocol formats of gpfpd, bdfpd, inspvasa, bdfpdb, rawimusb and INStest are given below, as shown in the following table.

Table 5 gpfpd format

Serial	Name	Meaning	Data type	Unit
	\$GPFPD	Format header	_	_
	GPSWeek	Current Week Number Since	Integer	_
	GPS cycles	GPS cycles per second	Floating-	S
	Yaw Angle	Yaw 0 ~ 360 degrees, clockwise	Floating-	Degr
	Pitch Angle	Pitch angle -90 ° ~ 90 °	Floating-	Degr

Serial	Name	Meaning	Data type	Unit
	Roll Angle	Roll angle -180 ° ~ 180 °	Floating-	Degr
	Latitude	Combined Output Latitude -90 ° ~	Floating-	Degr
	Longitude	Combined output longitude -180 °	Floating-	Degr
	Height	Height of the combined output	Floating-	m
	East speed	Combined output east speed	Floating-	m/s
	North	Combined output north speed	Floating-	m/s
	Sky speed	Combined output speed	Floating-	m/s
	Baseline	Distance between centers of two	Integer	Meter
	NSV1	Number of satellites for antenna 1	Integer	Α
	NSV2	Number of satellites for antenna 2	Integer	Α
	Satellite	Satellite status 0: unavailable, 1:	Integer	_
	Check code	Check code (value after exclusive or	Hexadeci	_
	<cr><lf></lf></cr>	Fix the tail of the package	_	_

Table 6 bdfpd format

Serial	Name	Meaning	Data	Unit
	\$BDFPD	Format header	_	
	GPSWeek	Current Week Number Since	Integer	_
	GPS cycles	GPS cycles per second	Floating-	S
	Yaw Angle	Yaw 0 ~ 360 degrees, clockwise	Floating-	Degre
	Pitch Angle	Pitch Angle Pitch angle -90 ° ~ 90 °		
	Roll Angle	Roll angle -180 ° ~ 180 °	Floating-	Degre
	Latitude	Combined Output Latitude -90 ° ~	Floating-	Degre
	Longitude	Combined output longitude -180 °	Floating-	Degre
	Height	Height of the combined output	Floating-	m
	East speed	Combined output east speed	Floating-	m/s
	North	Combined output north speed	Floating-	m/s
	Sky speed	Combined output speed	Floating-	m/s

Serial	Name	Meaning	Data	Unit
	NSV1	Number of satellites for antenna 1	Integer	Α
	NSV2	NSV2 Number of satellites for antenna 2		Α
	Positioning	Postype in bestpos, see Table 13	Integer	_
	Directional	The postype in heading is shown in	Integer	_
	Check code Check code (value after exclusive of		Hexadeci	_
	<cr><lf></lf></cr>	R> <lf> Fix the tail of the package</lf>		_

Table 7 Format of bdfpdl

Serial	Name	Meaning	Data	Unit
	\$BDFPD	Format header	_	_
	GPSWeek	Current Week Number Since	Integer	_
	GPS cycles	GPS cycles per second	Floating-	S
	Yaw Angle	Yaw 0 ~ 360 degrees, clockwise	Floating-	Degre
	Pitch Angle	Pitch angle -90 ° ~ 90 °	Floating-	Degre
	Roll Angle	Roll angle -180 ° ~ 180 °	Floating-	Degre
	Latitude	Combined Output Latitude -90 ° ~	Floating-	Degre
	Longitude	Combined output longitude -180 °	Floating-	Degre
	Height	Height of the combined output	Floating-	m
	East speed	Combined output east speed	Floating-	m/s
	North	Combined output north speed	Floating-	m/s
	Sky speed	Combined output speed	Floating-	m/s
	X-axis	IMU is on the right	Floating-	°/s
	Y-axis	Before the IMU system	Floating-	°/s
	Z-axis	Attach the IMU	Floating-	°/s
	X-axis	IMU is on the right	Floating-	m/s2
	Y-axis	Before the IMU system	Floating-	m/s2
	Z-axis	Attach the IMU	Floating-	m/s2
	NSV1	Number of satellites for antenna 1	Integer	Α

Serial	Name	Meaning	Data	Unit
	NSV2	Number of satellites for antenna 2	Integer	Α
	Positioning Postype in bestpos, see Table 13		Integer	_
	Directional	The postype in heading is shown in	Integer	_
	Check code	Check code (value after exclusive or	Hexadeci	_
	<cr><lf> Fix the tail of the package</lf></cr>		_	_

Table 8 Format of inspvasa

Serial	Name	Meaning	Data	Unit
	%INSPVASA	Format header	_	_
	GPSWeek	Current Week Number Since	Integer	_
	GPS cycles per	GPS cycles per second	Floating-	S
	GPSWeek	Current Week Number Since	Integer	_
	GPS cycles per	GPS cycles per second	Floating	S
	Latitude	Combined Output Latitude	Floating-	Degree
	Longitude	Combined output longitude	Floating-	Degree
	Height	Height of the combined	Floating-	m
	North speed	Combined output north	Floating-	m/s
	East speed	Combined output east speed	Floating-	m/s
	Sky speed	Combined output speed	Floating-	m/s
	Roll Angle	Roll angle -180 ° ~ 180 °	Floating-	Degree
	Pitch Angle	Pitch angle -90 ° ~ 90 °	Floating-	Degree
	Yaw Angle	Yaw 0 ~ 360 degrees,	Floating-	Degree
	INS status	See Table 11	_	_
	Check code	Check code (number	Hexadeci	_
	<cr><lf></lf></cr>	Fix the tail of the package	_	_

Table 9bdfpdb Protocol Description

Serial	Number	Definition	Meaning	Data type	Remark

Serial	Number	Definition	Meaning	Data type	Remark
	1	Frame	0хаа	_	
1	2		0x44	_	Header
	3	header	0x10	_	rieduei
2	4	Message	0x40	_	
3	5-8	Week of	Current Week	unsigned	_
4	9-16	Week	GPS cycles per	double	_
5	17-20	Yaw Angle	Yaw 0 ~ 360	float	_
6	21-24	Pitch Angle	Pitch angle -90 ° ~	float	_
7	25-28	Roll Angle	Roll angle -180 ° ~	float	_
8	29-36	Latitude	Combined Output	double	_
9	37-44	Longitude	Combined output	double	_
10	45-48	Height	Height of the	float	_
11	49-52	East speed	Combined output	float	_
12	53-56	North	Combined output	float	_
13	57-60	Sky speed	Combined output	float	_
14	61-62	NSV1	Number of	unsigned	_
15	63-64	NSV2	Number of	unsigned	_
16	65-66	Positioning	Postype in bestpos,	unsigned	_
17	67-68	Directional	The postype in	unsigned	_
18	69-72	Checksum	5-68 bytes 4-byte	_	_

Table 10 Description of rawimusb protocol

Seri	Numbe	Definition	Meaning	Data type	Remark
	1		Охаа	_	
1	2	Frame header	0x44	_	Hoodor
	3		0x13	_	Header
2	4	Message length	0x28	_	
3	5-6	Message ID	0x145	_	_

Seri	Numbe	Definition	Meaning	Data type	Remark
4	7-8	Week of GNSS		unsigned	_
5	9-12	Week second	ms	unsigned	_
6	13-16	Week of GNSS	_	unsigned	
7	17-24	Week second	S	double	
8	25-28	IMU status word	See Table 14	unsigned	
9	29-32	Z-direction	m/s2	int	200*200*2-3
10	33-36	-Y accelerometer	m/s2	int	200*200*2-3
11	37-40	X-direction	m/s2	int	200*200*2-3
12	41-44	Z-direction	°/s	int	200*720*2-3
13	45-48	-Y-gyro output	°/s	int	200*720*2-3
14	49-52	X-direction	°/s	int	200*720*2-3
15	53-56	Checksum	1-52 byte	unsigned	_

Table 11 INStest protocol description

Serial	Numb	Definition	Meaning	Data type	Remark
	1	Frame header	0x5a	_	- Header
	2	Frame neader	0x5a	_	пеацеі
	3	Data length	0x3c	unsigned	_
	4-7	Week second	S	float	_
	8-9	Heading angle	o	unsigned	0.01
	10-11	Pitch Angle	o	short	0.01
	12-13	Roll Angle	o	short	0.01
	14-15	North speed	m/s	short	0.01
	16-17	East speed	m/s	short	0.01
	18-19	Sky speed	m/s	short	0.01
	20-23	Longitude	0	int	0.0000001
	24-27	Latitude	0	int	0.0000001
	28-29	Height	m	short	0.1

Serial	Numb	Definition	Meaning	Data type	Remark
	30-31	100 C serial		unsigned	_
	32.0	Com1status		_	_
	32.1	Com2 status		_	_
	32.2	Com3 status		_	_
	32.3	GNSS		_	_
	32.4	Com4 status	0: normal, 1:	_	_
	32.5	GNSS com3	fault	_	_
	32.6	Status of the can		_	_
	32.7	Odometer		_	_
	33.0	PPS status		_	_
	33.1	EVENT status		_	_
	34	INS status word	0 x00:	_	_
	35	GNSS	Postype in	_	_
	36	Number of	_	_	_
	37	Dual antenna	The postype	_	_
	38	Position the	_	_	_
	39-40	Gyro X axis	°/s	short	0.01
	41-42	Gyro Y-axis	°/s	short	0.01
	43-44	Gyro Z axis	°/s	short	0.01
	45-46	Add Table X Axis	g	short	0.001
	47-48	Add table Y axis	g	short	0.001
	49-50	Add table Z axis	g	short	0.001
	51	IMU fault word	See Table 15		
	52-59	Spare	_	_	_
	60-63	Frame count	0x0000~0xFF	unsigned	_
	64	Checksum	3-byte to	_	_

INS status word	Status word description
INS_INACTIVE	IMU logs are present, but the alignment routine
INS ALIGNING	INS is in alignment mode.
INS_SOLUTION_GOOD	The INS filter is in navigation mode and the INS

Table 13 postype description

Туре	Type definition	Type description			
numeric					
value					
0	NONE	No solution			
1	FIXEDPOS	Position has been fixed by the FIX POSITION			
		command			
2	FIXEDHEIGHT	Position has been fixed by the FIX HEIGHT/AUTO			
		command			
8	DOPPLER_VELOCI	Velocity computed using instantaneous Doppler			
	TY				
16	SINGLE	Single point position			
17	PSRDIFF	Pseudorange differential solution			
18	WAAS	Solution calculated using corrections from an			
		WAAS			
19	PROPAGATED	Propagated by a Kalman filter without new			
		observations			
20	OMNISTAR	OmniSTAR VBS position			

32	L1_FLOAT	Floating L1 ambiguity solution	
33	IONOFREE_FLOAT	Floating ionospheric-free ambiguity solution	
34	NARROW_FLOAT	Floating narrow-lane ambiguity solution	
48	L1_INT	Integer L1 ambiguity solution	
50	NARROW_INT	Integer narrow-lane ambiguity solution	
64	OMNISTAR_HP	OmniSTAR HP position	
65	OMNISTAR_XP	OmniSTAR XP or G2 position	
68	PPP_CONVERGIN	Converging PPP solution	
	G		
69	PPP	Converged PPP solution	
70	OPERATIONAL	Solution accuracy is within UAL operational limit	
71	WARNING	Solution accuracy is outside UAL operational limit	
		but within warning limit	
72	OUT_OF_BOUNDS	Solution accuracy is outside UAL limits	

Table 14 IMU Status Word Description

Bit	Type description	
0	X Gyro status	
1	Y Gyro status	1: normal, 0: fault
2	Z gyro status	
3	Spare	
4	X Accelerometer Status	1: normal, 0: fault
5	Y Accelerometer Status	1. Hoffilal, 0. fault

Bit	Type description	
6	Z Accelerometer Status	
7-31	Spare	_

Table 15 IMU Fault Word Description

Bit	Type description	
0	X Gyro status	
1	Y Gyro status	0: normal, 1: fault
2	Z gyro status	
3	X Accelerometer Status	
4	Y Accelerometer Status	0: normal, 1: fault
5	Z Accelerometer Status	
6	Spare	_
7	Spare	_

4.3.2.4.332 bit CRC check calculation method

The 32-bit CRC check calculation method can be obtained by using the following C language function.

```
#define CRC32 POLYNOMIAL 0xEDB88320L
Calculate a CRC value to be used by CRC calculation functions.
unsigned long CRC32Value(int i) {
    int i:
    unsigned long ulCRC;
    ulCRC = i;
    for (j = 8; j > 0; j--) {
         if ( ulCRC & 1 )
             ulCRC = ( ulCRC >> 1 ) ^ CRC32_POLYNOMIAL;
             ulCRC >>= 1;
    return ulCRC;
/* -----
Calculates the CRC-32 of a block of data all at once
ulCount - Number of bytes in the data block
ucBuffer - Data block
unsigned long CalculateBlockCRC32( unsigned long ulCount, unsigned char
    unsigned long ulTemp1;
    unsigned long ulTemp2;
    unsigned long ulCRC = 0;
    while ( ulCount-- != 0 ) {
         ulTemp1 = ( ulCRC >> 8 ) & 0 \times 000  FFFFFFL;
         ulTemp2 = CRC32Value( ((int) ulCRC ^ *ucBuffer++ ) & 0xFF );
         ulCRC = ulTemp1 ^ ulTemp2;
    return( ulCRC );
```

4.3.2.5 initial value configuration

Initial longitude and latitude configuration, configuration instructions are:

initialpos LONGITUDE LATILUDE

Where LONGITUDE and LATITUDE are configured local longitude and latitude values in degrees.

4.3.2.6 function module configuration

Functional modules with open configuration mainly include zero velocity correction and output position smoothing.

4.3.2.6.1 "Zero Velocity Trim" Configuration

The zero-velocity correction function mainly means that the integrated navigation

system detects the sensitive information, and if the integrated navigation system is judged to be zero-velocity, the corresponding correction is carried out.

In the integrated navigation process of this product, the "zero velocity correction" is enabled by default. If the satellite information is invalid for a long time in the integrated navigation state, and the user wants to get the pure inertial navigation information, it is recommended to close the zero velocity correction mode.

The zero speed correction configuration instructions are as follows:

inszupt switch

The switch value is either disable or enable, where disable turns the feature off and enable turns the feature on.

4.3.2.6.2 Position Output Smoothing configuration

In order to get more smooth position information, the navigation software adds the function of position output smoothing, which makes the position noise smaller after smoothing. (Note: "Position Output Smoothing" shall be closed in RTK state) In the integrated navigation process of this product, "Position Output Smoothing" is off by default. In order to facilitate the user's selection, this function can be configured. The configuration instructions are as follows:

possmooth switch

The switch value is either disable or enable, where disable turns the feature off and enable turns the feature on.

4.3.2.7 carrier type configuration

According to different carriers installed in the integrated navigation system, the

carrier type configuration is required, and different algorithm processing is carried out in the integrated navigation system according to different carrier types.

The configuration instructions are as follows:

carrier vehicle/ship/air

They are vehicle-mounted, ship-mounted and airborne in turn.

After the configuration is completed, you need to enter the save command "saveconfig", and then hard start or enter the "# reset" command. The carrier type configuration will be valid after startup. The integrated navigation system does not support current configuration and current use during use, and must be restarted. After the carrier type is configured as the vehicle-mounted type, the integrated navigation system is required to be installed and fixed on the vehicle, and the heading of the integrated navigation system is consistent with the head direction of the vehicle, with an error of not more than 10 degrees.

4.3.2.8 GNSS antenna mast arm configuration

According to the relative installation relationship between the antenna and the integrated navigation system, it is necessary to configure the antenna rod arm. The lever arm value between the integrated navigation system and the antenna must be accurate to millimeter (mm) during measurement, especially during RTK operation. Any lever arm measurement error will directly enter the position error output by the integrated navigation system. During installation and use, the integrated navigation system should be as close as possible to the main antenna, especially in the horizontal position. The command is required to be completed before or during

the alignment of the integrated navigation system on the stationary base and before the alignment of the integrated navigation system on the moving base.

Once the configuration is complete, it needs to be saved via "saveconfig".

The configurations include a master antenna rod arm configuration and a slave antenna rod arm configuration.

The main antenna configuration instructions are as follows:

setimutoantoffsetarmX armY armZ

The slave antenna configuration instructions are as follows:

setimutoantoffset2 armX armY armZ

Where armX, armY and armZ are the configured lever arm values in meters, representing the components of the vector from the integrated navigation system to the antenna phase center in the integrated navigation system carrier coordinate system, and the integrated navigation system carrier coordinate system is selected as the right front top (XYZ). For the example in Figure 5, armX and armY should be negative, and armZ should be positive.

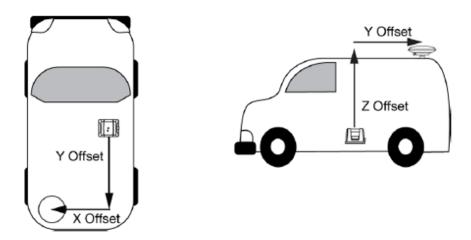


Figure 5 Schematic diagram of antenna rod arm

4.3.2.9 Output Lever Arm Settings

The default value for the product output lever arm configuration is [0,0,0] (upper right front), which outputs the position and speed values at the integrated navigation system. If the position and speed of the user's test point need to be output, the output lever arm should be set according to the relative installation relationship between the test point and the integrated navigation system.

The lever arm value from the configuration of integrated navigation system to the test point must be accurate to millimeters (mm) during measurement, especially during RTK operation, any lever arm measurement error will directly enter the position error output by the integrated navigation system. The command is required to be completed before or during the alignment of the integrated navigation system on the stationary base and before the alignment of the integrated navigation system on the moving base. Once the configuration is complete, it needs to be saved via "saveconfig".

The output lever arm configuration commands are as follows:

setimutosensoroffsetarmX armY armZ

Where armX, armY, and armZ are the configured lever arm values, in meters, representing the components of the vector from the integrated navigation system to the test point in the integrated navigation system carrier coordinate system, and the integrated navigation system carrier coordinate system is selected as the right front top (XYZ). For the example in Figure 6, armY and armZ should be positive.

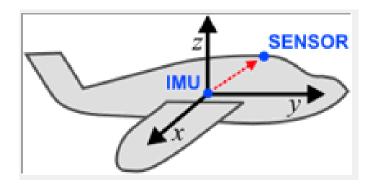


Fig. 6 Schematic diagram of output lever arm

Setting of mounting angle of 4.3.2.9

The attitude and heading information output by the product are Euler angles of the product coordinate system relative to the geographic coordinate system. The angle installation relationship between the product and the carrier coordinate system is the installation angle, and the default configuration value is [0,0,0] (pitch, heading, roll), that is, the product coordinate system and the installation carrier coordinate system coincide. If there is an installation angle when the product is installed on the carrier, and the Euler angle of the carrier coordinate system relative to the geographic coordinate system needs to be output by the product, the installation angle should be set according to the relative installation relationship between the product and the carrier.

Mounting angle configuration instructions are as follows:

vehiclebodyrotation angleX angleZ angleY

Where angleX, angleZ and angleY are the configured installation angle values, in degrees, representing the angles from the carrier coordinate system to the integrated navigation system coordinate system, in the order of pitch, course and roll.

4.3.2.10 output angle setting

The attitude and heading information output by the product are Euler angles of the product coordinate system relative to the geographic coordinate system. The angular relationship between the product and the specified coordinate system is the output angle, and the configuration default is [0,0,0] (heading, pitch, roll), that is, the product coordinate system is considered to coincide with the specified coordinate system. If the Euler angle of the specified coordinate system relative to the geographic coordinate system needs to be output, the output angle should be set according to the relative relationship between the product and the specified coordinate system.

The output angle configuration instructions are as follows:

outputattituderotationangleZ angleX angleY

Where angleZ, angleX, and angleY are the configured output angle values, in degrees, representing the angle from the specified coordinate system to the product coordinate system, in the order of heading, pitch, and roll.

Note: This function cannot be used in conjunction with the mounting angle setting.

4.3.2.11 forced rotation inertial navigation

When the integrated navigation system is in the integrated navigation state, the integrated navigation system can receive the forced rotation inertial navigation instruction and switch to the inertial navigation state. In this state, the integrated navigation system still receives the satellite navigation information for protocol transmission, but does not use the satellite navigation information to participate in

the integrated navigation calculation. After receiving the effective forced rotation inertial navigation command, the integrated navigation system feeds back the "cmd OK" "through the COM1 port.

Forced-turn inertial navigation commands are as follows:

#moddgitoins

4.3.3 system maintenance

4.3.3.1 firmware upgrade

When a firmware upgrade is required, proceed as follows:

Connect the power line and communication line, connect the COM1 port to the computer, and set the baud rate to 115200 bps;

When the system is powered on, the serial port tool interface displays the startup prompt information. Before the interface displays "10 9 8 7 6 5 4 3 2 1" to 1, send ":" (small colon, cancel the option of sending a new line) to the serial port, and the interface displays the updata flash information;

Select the firmware (generally *.bin2 file) to be upgraded through the serial port tool and send it;

After the sending is completed, the program automatically reloads and starts, enters the start prompt information, and starts normally;

The firmware upgrade is complete.

4.3.3.2 parameter upload

In general, the user does not need to upload the calibration parameters, and the configuration has been completed before leaving the factory. Under special

circumstances, if the user is required to upload and maintain the parameters, the following steps shall be followed:

After the system completes the startup prompt information normally, you can query the corresponding system number through the "log bdlist"/ "log rxstatus"; Send the "# modbd" command to the integrated navigation system through the COM1 port, and upload the "*.txt" calibration parameter file through the serial port after the "cmd OK" "is returned;

After the interface returns to the calibration parameter information, send the "# saveconfig"/ "saveconfig" command to save the parameters, and then reset the system by soft and hard reset to work normally.

5. Precautions

The main considerations are as follows:

The power-on and power-off time interval of the integrated navigation system shall not be less than 30 s, otherwise it is easy to cause damage to the inertial devices; Handle with care during handling, installation and use to avoid bumping, falling and bumping;

Do not change the output and baud rate configuration of the COM2 port of the satellite board.

Appendix

6.1 Satellite Receiver COM2 Output Configuration

The integrated navigation system receives satellite navigation information through

the COM2 port of the built-in satellite receiver. The COM2 port configuration command of the satellite receiver is as follows:

unlogall com2

serialconfig com2 115200

log com2 gprmc ontime 1

log com2 gpgga ontime 1

log com2 bestposa ontime 1

Log com2 rangecmpb ontime 1 (Note: this command is invalid without the original

data board)

log com2 bestposb ontime 0.2

log com2 headingb onchanged

log com2 bestvelb ontime 0.2

log com2 psrdopb ontime 1

dualantennaalign enable 5 1

frequencyout enablesync 100000 1000000000

saveconfig

Description of the differential configuration of the 6.2

The integrated navigation system can receive the differential correction information sent by the reference station through the communication link, work in the differential state, and achieve the positioning accuracy of centimeter level. Differential configuration mainly includes three parts: 1. Reference station setting; 2. Communication link setting; 3, mobile station settings. The data link is shown in

the following illustration.

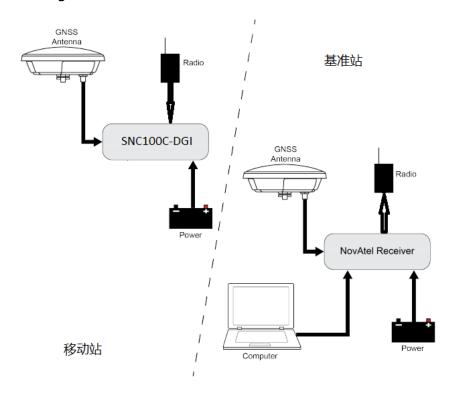


Figure 7 Data Link Diagram

Setting of 6.2.1 reference station

The differential reference station is the satellite receiver with the antenna of the satellite receiver installed in a fixed position. During the use of the satellite receiver, it is necessary to ensure that the antenna is fixed. In the working process of the differential reference station, the precise coordinates of the fixed position and the received satellite information are sent to the mobile station (the point to be positioned) through the communication link, which is used for the mobile station to carry out differential positioning calculation, realize differential high-precision positioning, and achieve centimeter-level positioning accuracy.

The NovAtel-718D satellite receiver card can be applied to the reference station, and it can be configured as the reference station mode. The specific configuration

instructions are as follows. See the NovAtel-718D user manual for details.

Table 16 Reference Station Configuration Instructions

Serial number	Instruction	Explain		
	fix position39.8122 116.1515 60.5	Set the known precise coordinates (latitude 39.8122, 经度 116.1515, altitude 60.5) as the reference station coordinate values		
1	posaveon 0.01 1.5 2.5	Autonomous positioning of the receiver 0.01 H; Or when the horizontal positioning standard deviation is less than or equal to 1.5 m and the height positioning standard deviation is less than or equal to 2.5 m, the average value of the positioning is taken as the coordinate value of the reference station		
2	serialconfig com1 9600	Set the output baud rate of the output interface COM1 of the reference station to 9600 bps		

Serial number	Instruction	Explain		
3	interfacemode com1 novatel rtcmv3 on	Configure COM1 input data type as novatel, output data type as rtcmv3, and enable command feedback		
4	log com1rtcm1075 ontime 1	GPS differential message		
5	log com1rtcm1125 ontime 1	BDS differential message		
6	log com1rtcm1085 ontime 1	GLO differential message		
7	log com1rtcm1033 ontime 10	Description of receiver and antenna		
8	log com1rtcm1005 ontime 10	Antenna reference point coordinates of RTK reference station		
9	saveconfig	Save the configuration		

6.2.2 communication link settings

The communication link can use 4G DTU or data radio, and the coverage of 4G DTU is the coverage of network information; Limited by national laws, the coverage of the digital radio station is about 10km. For the setting of 4G DTU and digital radio, please refer to the user manual of the corresponding product.

6.2.3 mobile station settings

The integrated navigation system is a differential mobile station. The mobile station receives the differential correction information of the reference station in real time, and receives the satellite signal to calculate the differential positioning, so as to realize the differential high-precision positioning. The integrated navigation system supports RTCM and RTCMV3 standard data protocols. GNSS COM1 or GNSS COM3 of the integrated navigation system can be configured as the receiving interface of differential correction information. The specific configuration instructions are as follows. See the NovAtel-718D user manual for details.

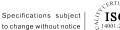
Table 17 Mobile station configuration command

Serial number	Instruction	Explain
1	serialconfig com1 9600	Set the input baud rate of the mobile station input interface COM1 to 9600 bps
2	interfacemode com1 rtcmv3 novatel on	Configure COM1 output data type as novatel, input data type as rtcmv3, and enable command feedback
3	saveconfig	Save the configuration

7. Update records

Seri	Versio	Change	Before	After the change	Reason	Changed by

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1	V1.00	20190605			New	Zhang
2	V1.01	20190616		Add appearance		Zhang
3	V1.02	20190814		Add technical indicators Add Output Lever Arm		Zhang Xiangliang
4	V1.03	20191125		Add bdpdb protocol		Zhang
5	V1.04	20191202		Add output protocol		Zhang
				description		Xiangliang
6	V1.05	20191216		Added description of		Zhang
7	V1.06	20191217		Added Pure Inertia Test		Zhang
8	V1.07	20200227	Combi	0.1°		Zhang
9	V1.08	20200229		Add opening position smooth caution		Zhang Xiangliang
10	V1.09	20200305	Postpr	0.02°		Zhang
11	V1.10	20200318	RTK	2cm+1ppm		Zhang
12	V1.11	20200325		Increase GNSS failure horizontal position		Zhang Xiangliang
13	V1.12	20200723		Increase the output		Zhang
14	V1.13	20200804		Add protocol	Perfect	Zhang
15	V1.14	20230411		New Structure Status	Update	YHS
16	V1.15	20230424		Bdfpdb protocol	Update	YHS
17	V1.16	20230514		Bdfpdl protocol update	Update	YHS